

COMBINED LOADINGS (CHỊU LỰC PHỨC TẠP)

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I/ INTRODUCTION

- Biaxial Flexure (or Unsymmetrical Bending)
- Traction (Compression) & Bending
- Torsion and bending
- General loadings



I/ INTRODUCTION (cont.)

- PRINCIPLE OF SUPERPOSITION

“The internal forces, stresses, strains, displacements,..., due to multiple causes are the sum of these quantities due to each one acting independently”

$$S(P_1, P_2, \dots, P_n, \Delta, \dots) = S(P_1) + S(P_2) + \dots + S(P_n) + \dots + S(\Delta) + \dots$$



I/ INTRODUCTION (cont.)

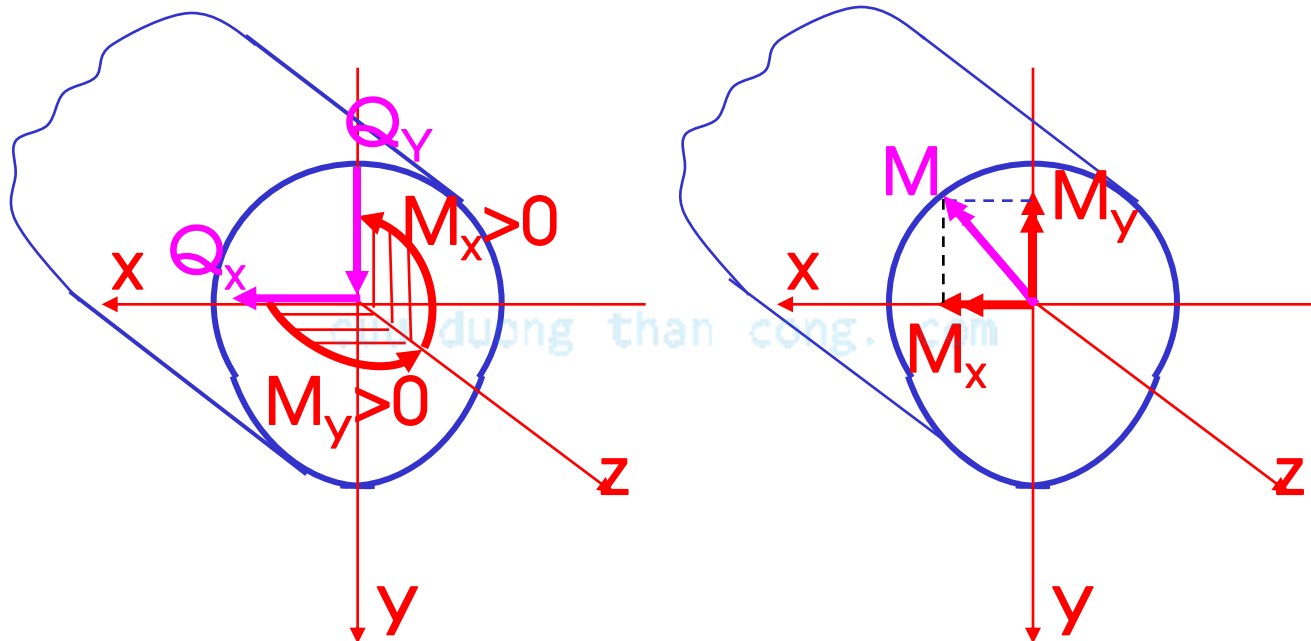
- Condition of application (Điều kiện áp dụng)
 - ✓ Material is linearly elastic
 - ✓ Small displacements

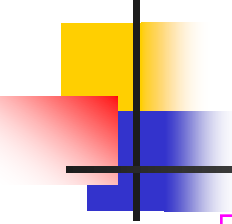
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II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING)

- **Definition:** when there are two bending moments and two shear forces in two principal planes M_x, M_y, Q_x, Q_y

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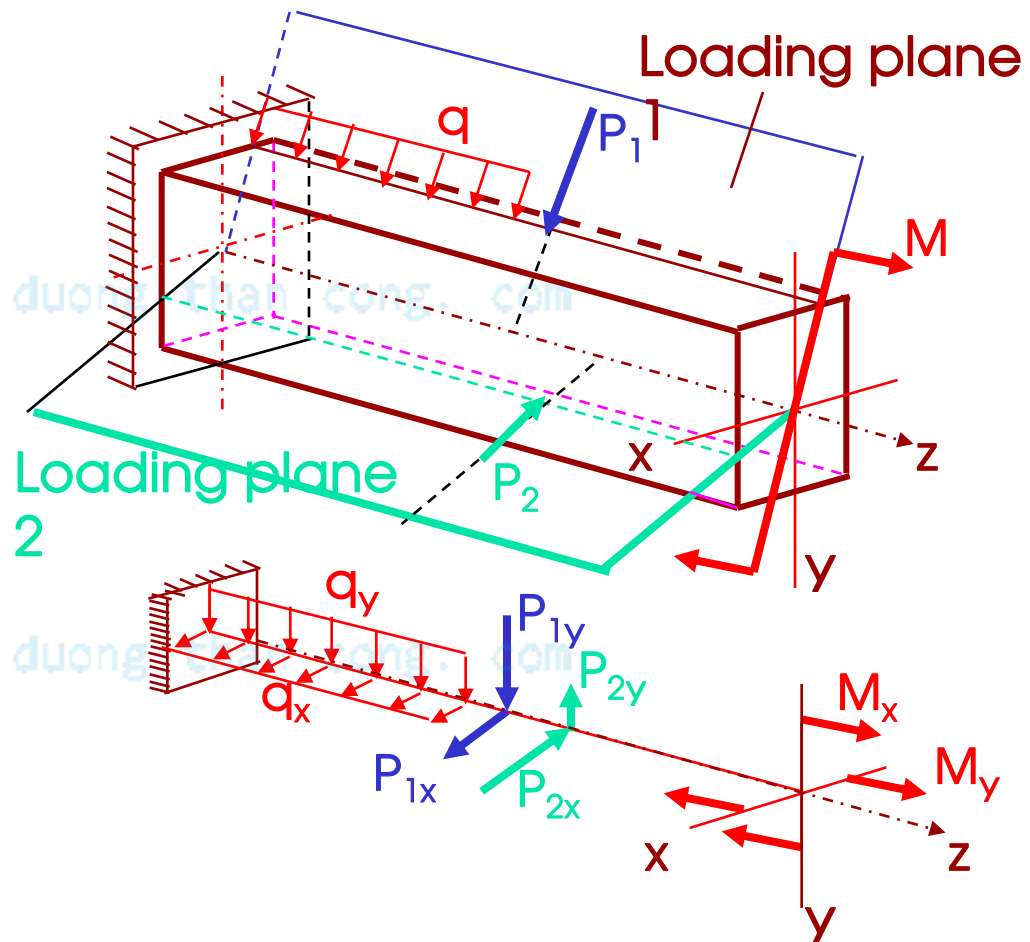
II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.)

- **Notations:**

- ✓ Bending moments are positive if they have tendency to extend the corresponding positive fibers
- ✓ Expressing bending moment by a two-arrow vector → the resulting moment M of M_x and M_y lies on an arbitrary plane called loading plane (mặt ph. tải trọng)

II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.)

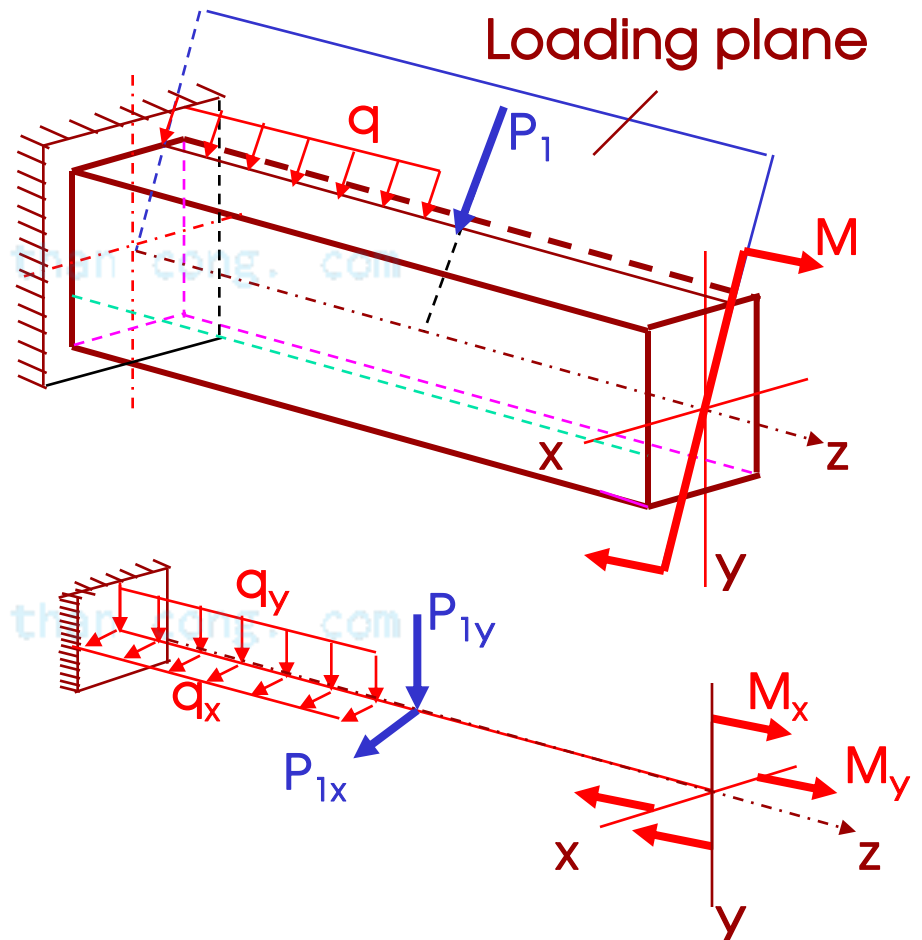
a/ Loadings are in two principal planes



II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.)

b/ Loadings are in 1 arbitrary plane containing its axis

→ Unsymmetrical Bending

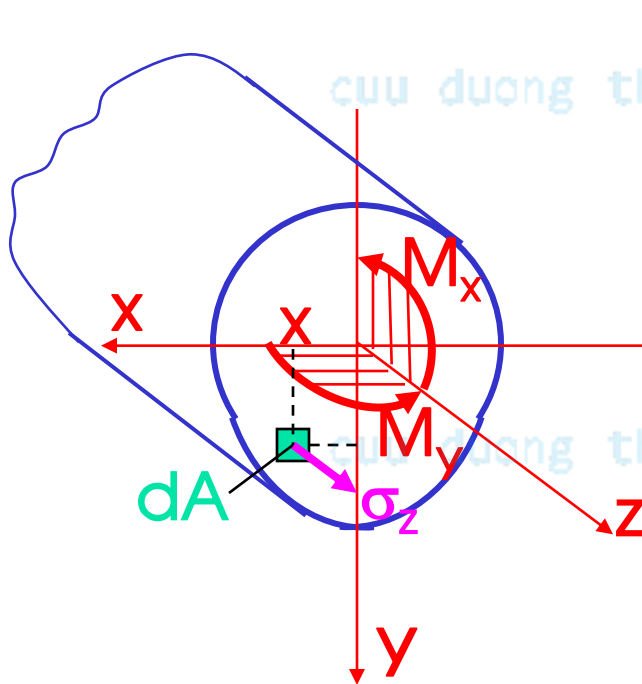


II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.)

- Stress at a point (x,y): Normal Stress, σ_z

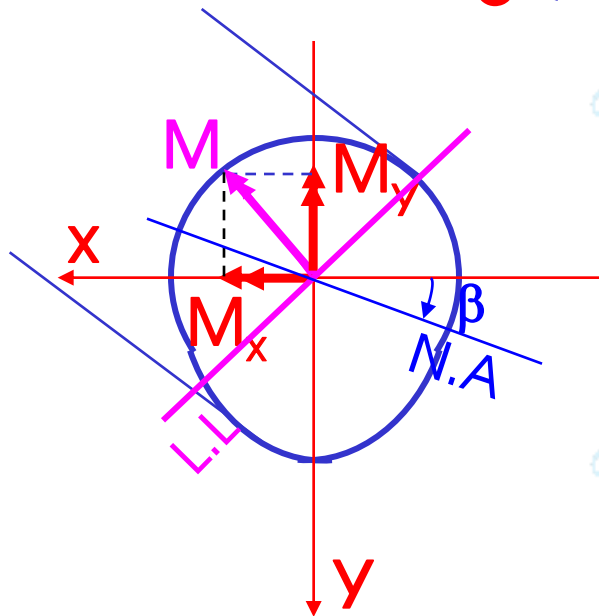
$$\sigma_z = \frac{M_x}{I_x} y + \frac{M_y}{I_y} x$$

Or (hay)

$$\sigma_z = \pm \frac{|M_x|}{I_x} |y| \pm \frac{|M_y|}{I_y} |x|$$


II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.)

- Neutral Axis (Trục trung hòa): set of point (x,y) having (tập hợp các điểm (x,y) có) $\sigma_z(x,y) = 0$



$$\sigma_z = \frac{M_x}{I_x} y + \frac{M_y}{I_y} x = 0 \rightarrow y = ax$$

$$\text{with } a = \tan \beta = -\frac{M_y}{M_x} \cdot \frac{I_x}{I_y} = -\frac{1}{\tan \alpha} \cdot \frac{I_x}{I_y} :$$

slope of N.A.

$$\text{where } \tan \alpha = \frac{M_x}{M_y} : \text{slope of load line}$$



II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) – Normal Stress

- **Properties:** $\tan \alpha . \tan \beta = - \frac{I_x}{I_y}$

✓ In general, $I_x \neq I_y$: $\tan \alpha . \tan \beta \neq -1$

→ load line (L.L.) is not $\perp$ N.A.

✓ If $I_x = I_y$ (square, circular sections,...):

$$\tan \alpha . \tan \beta = -1$$

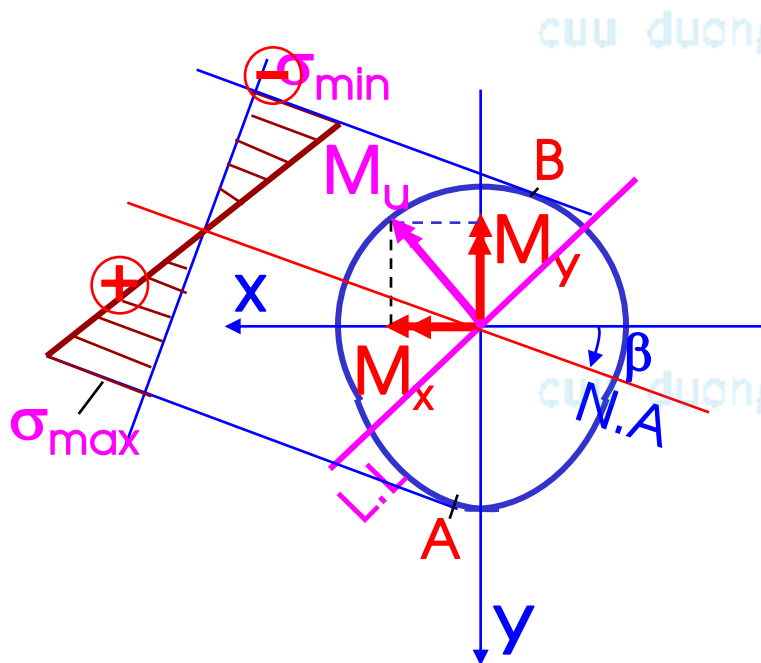
→ all axes become the principal axes

→ uniplanar bending under $M_u = \sqrt{M_x^2 + M_y^2}$

II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) – Normal Stress

■ Normal Stress Distribution

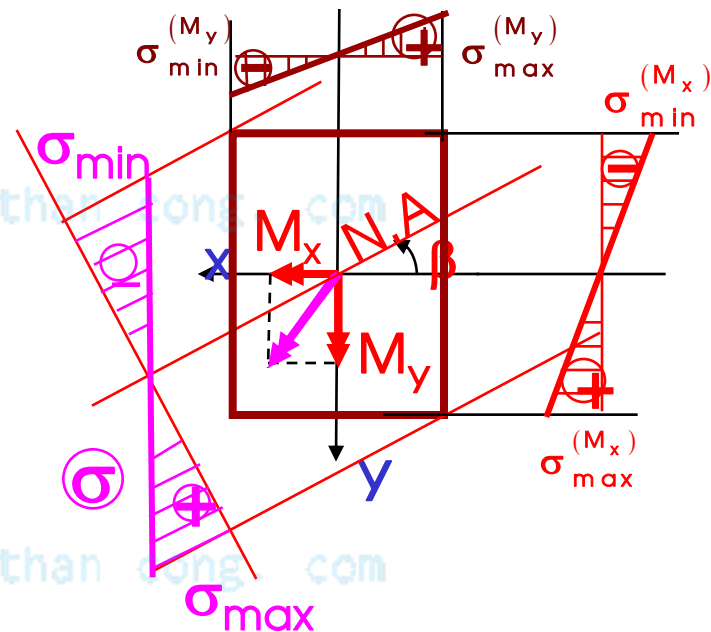
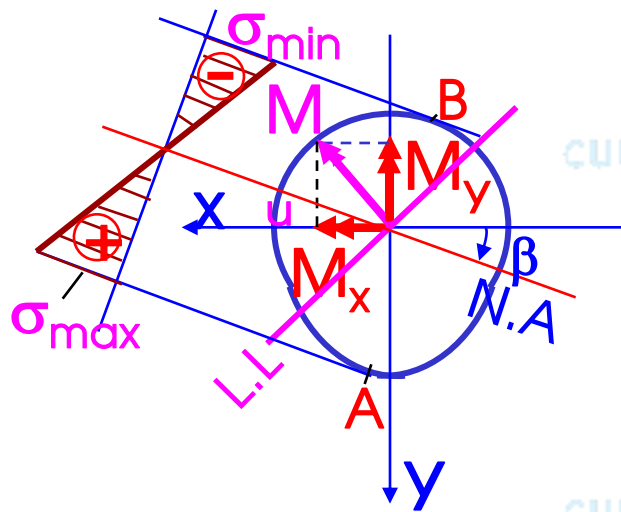
✓ Arbitrary uniaxially symmetric section



$$|\sigma_{\max}| = |\sigma_A| = \frac{|M_x|}{I_x} |y_A| + \frac{|M_y|}{I_y} |x_A|$$

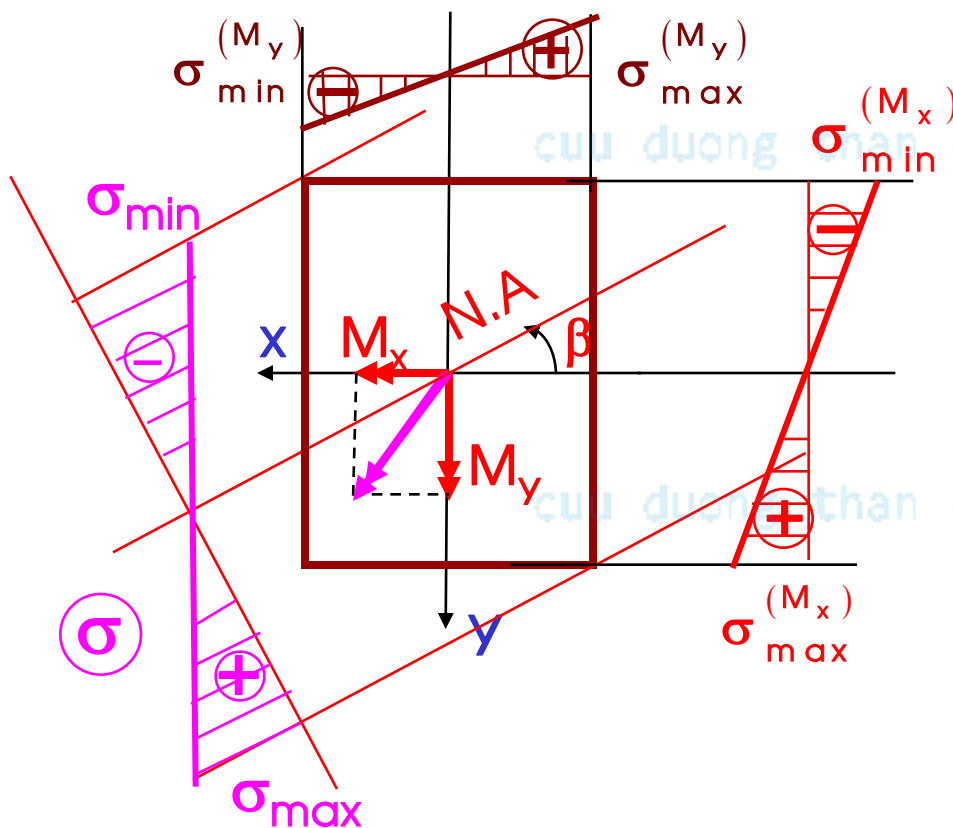
$$|\sigma_{\min}| = |\sigma_B| = \frac{|M_x|}{I_x} |y_B| + \frac{|M_y|}{I_y} |x_B|$$

II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) – Normal Stress



II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) – Normal Stress

■ Biaxially symmetric sections



$$\sigma_{\max} = + \frac{|M_x|}{W_x} + \frac{|M_y|}{W_y}$$

$$\sigma_{\min} = - \frac{|M_x|}{W_x} - \frac{|M_y|}{W_y}$$

with $W_x = \frac{I_x}{y_{\max}}$

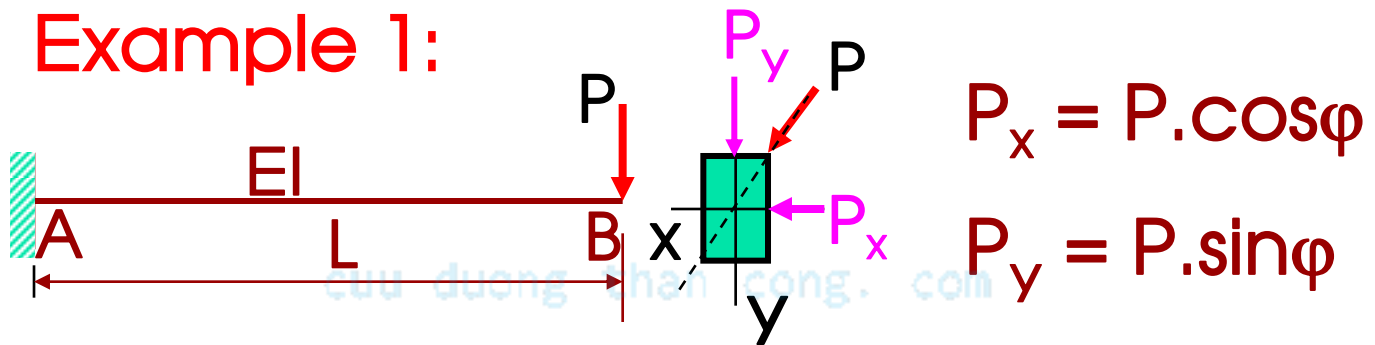
$$W_y = \frac{I_y}{x_{\max}}$$

II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) – Displacements

- Displacements in Unsymmetric Bending

$$\vec{f} = \vec{f}_x + \vec{f}_y \quad \text{with} \quad f = \sqrt{f_x^2 + f_y^2}$$

- Example 1:



Compute the deflection of point B?



II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) – Displacements

- ✓ Deflection of point B only due to P_y

$$f_y = \frac{P_y L^3}{3EI_x} = \frac{(P \sin \phi) L^3}{3EI_x}$$

- ✓ Deflection of point B only due to P_x

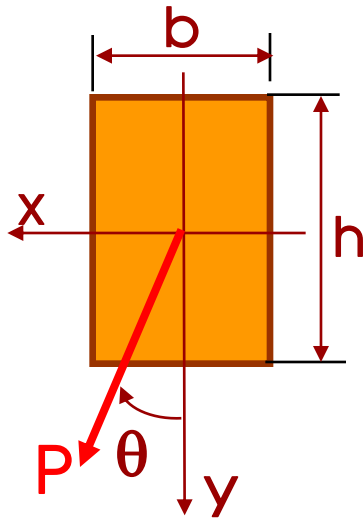
$$f_x = \frac{P_x L^3}{3EI_y} = \frac{(P \cos \phi) L^3}{3EI_y}$$

- ✓ Total deflection of the free end B

$$f = \sqrt{f_x^2 + f_y^2}$$

II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) –

■ Example 2:



✓ A wood of cantiliver beam of rectangular cross section supports a inclined load P at its free end. Calculate the maximum tensile stress $\sigma_{\max}$ due to the load P ?

✓ Data for the beam:

$b = 75 \text{ mm}$, $h = 150 \text{ mm}$, $L = 1.4 \text{ m}$,
 $P = 800 \text{ N}$, and $\theta = 30^\circ$

II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) – Example 2

■ Solution:

✓ Bending moments at the fixed end section:

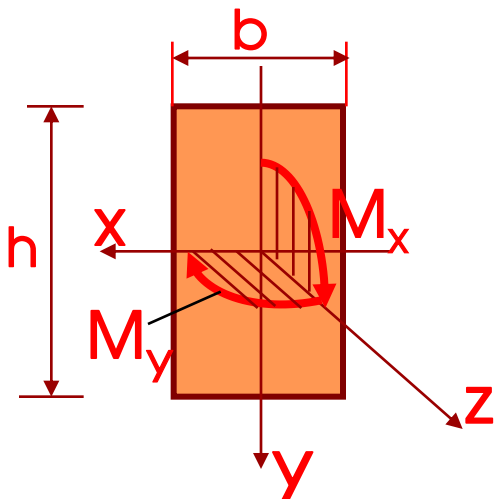
$$M_x = -800 \times (1.4) \times \cos 30^\circ = -969.95 \text{ N.m}$$

$$M_y = -800 \times (1.4) \times \sin 30^\circ = -560 \text{ N.m}$$

✓ Geometrical properties:

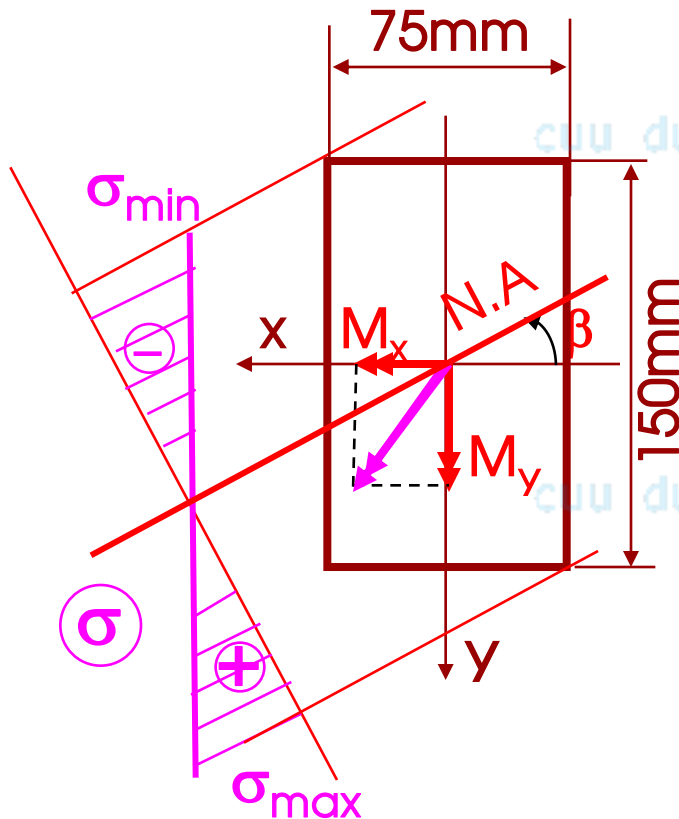
$$W_x = \frac{bh^2}{6} = \frac{(0.075)(0.15)^2}{6} = 2.8 \times 10^{-4} \text{ m}^3$$

$$W_y = \frac{hb^2}{6} = \frac{(0.15)(0.075)^2}{6} = 1.4 \times 10^{-4} \text{ m}^3$$



II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) – Example 2

✓ Maximum Stress



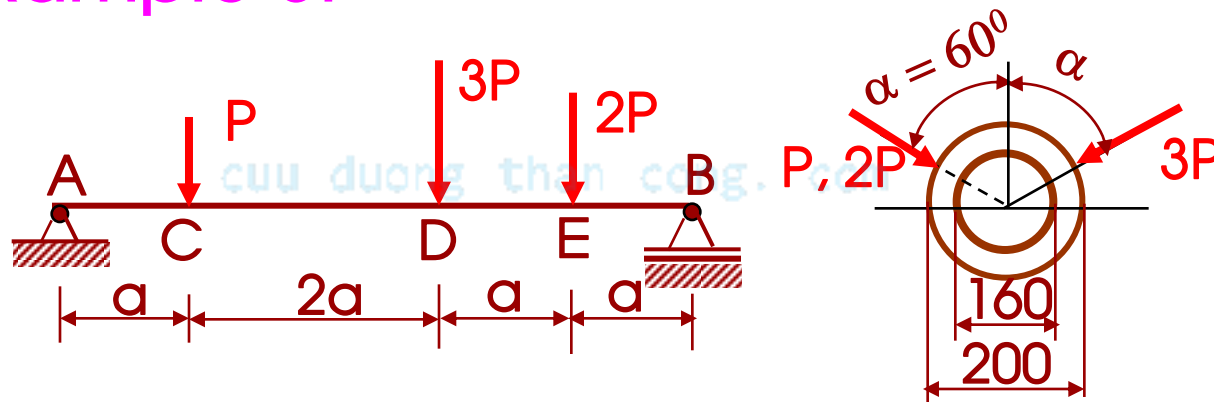
$$\sigma_{max} = \frac{969.95}{2.8 \times 10^{-4}} + \frac{560}{1.4 \times 10^{-4}}$$

$$= 7.46 \times 10^6 \text{ Pa} = 7.46 \text{ MPa}$$

$$\sigma_{min} = -\sigma_{max} = -7.46 \text{ MPa}$$

II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) – Example 3

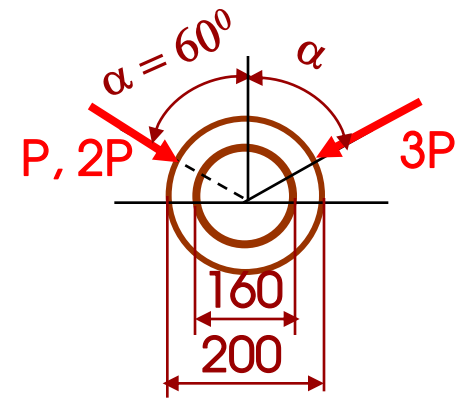
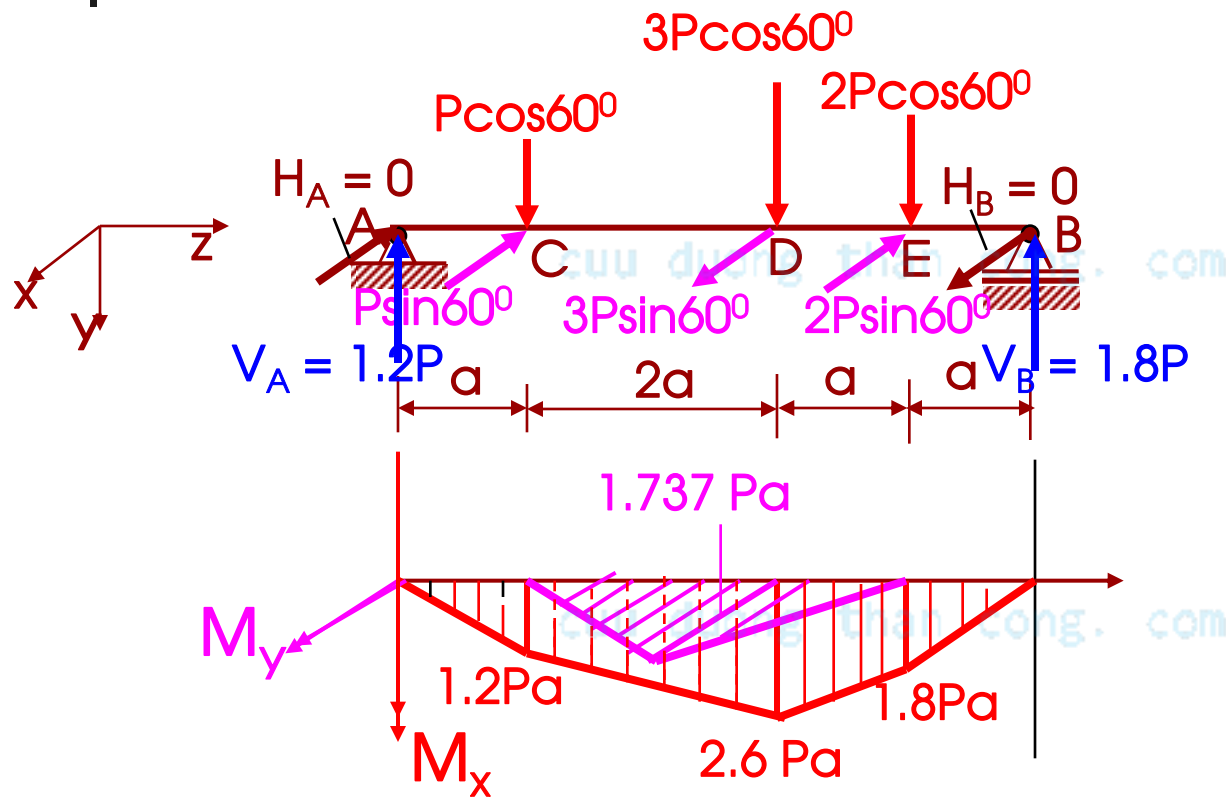
■ Example 3:



$$[\sigma] = 160 \text{ MPa}; \quad a = 0.5 \text{ m}$$

- Calculate the allowable load value [P]
- Determine the N.A. position

II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) – Example 3



II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) – Example 3

■ Solution

✓ Critical section at D:

$$\left. \begin{array}{l} M_x = 2.6 P a, \\ M_y = \sqrt{3} P a \end{array} \right\} \rightarrow M_u = \sqrt{M_x^2 + M_y^2} = 3.124 P a$$

✓ Strength condition:

$$\frac{M_u}{W_u} = \frac{5.655 P a}{\frac{\pi}{32} D^3 \left[1 - \left(\frac{d}{D} \right)^4 \right]} = \frac{5.655 \times 0.5 \times P}{\frac{\pi}{32} \times 0.2^3 \left[1 - \left(\frac{0.16}{0.2} \right)^4 \right]} \leq 160 \text{ MPa}$$

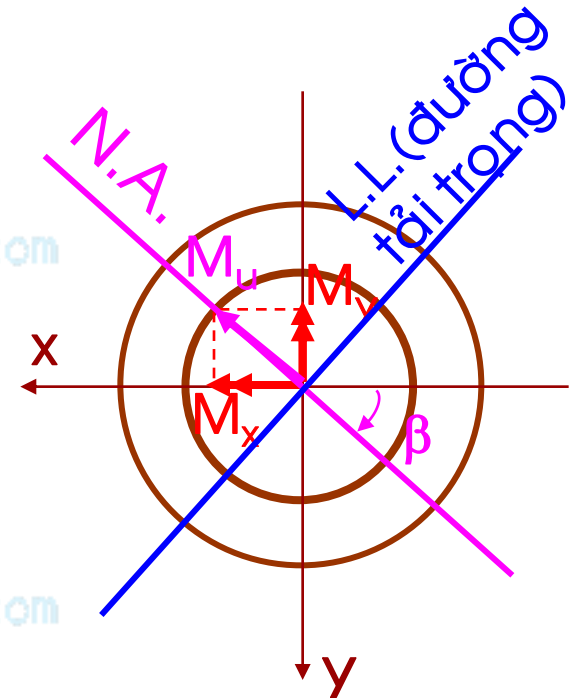
$$\rightarrow P \leq 0.0475 \text{ MN} = 47.5 \text{ kN}$$

II/ BIAXIAL FLEXURE (UNSYMMETRICAL BENDING) (cont.) – Example 3

✓ Neutral Axis

$$\tan \beta = -\frac{M_y}{M_x} \cdot \frac{I_x}{I_y} = -\frac{\sqrt{3}}{2.6} = -0.666$$

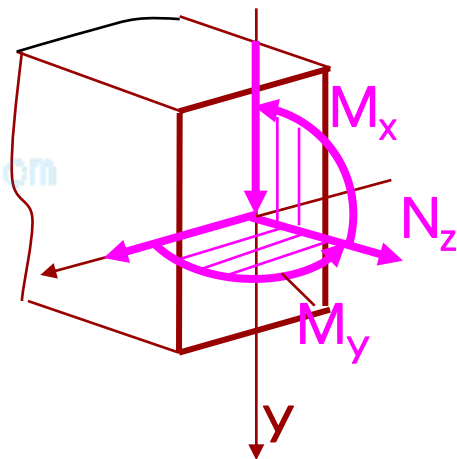
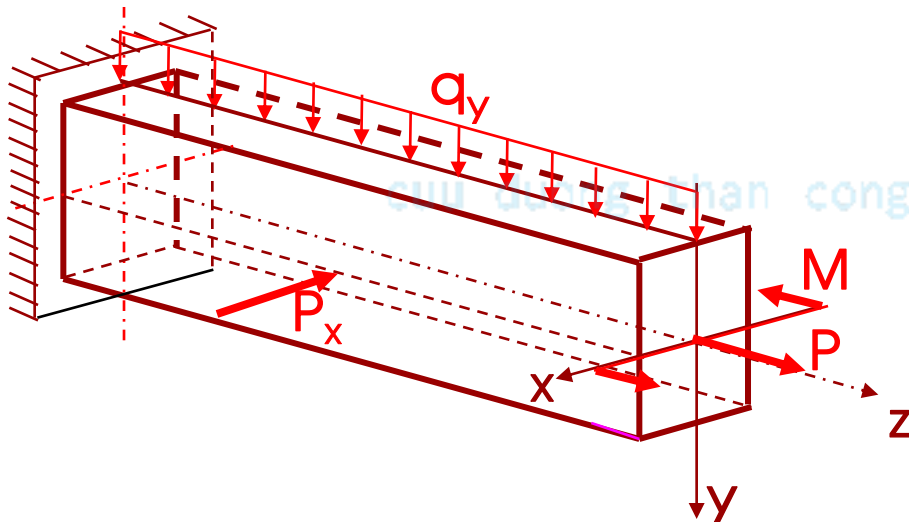
$$\rightarrow \beta = -33,6^\circ$$



III/ COMBINED BENDING AND TENSION OR COMPRESSION (UỐN VÀ KÉO HOẶC NÉN)

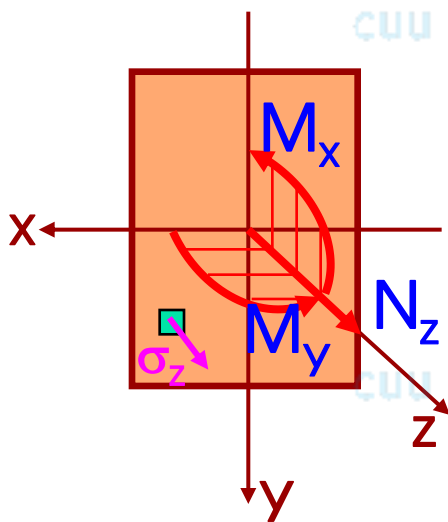
- Internal forces:

- ✓ Normal force: N_z
- ✓ Bending moment(s): M_x and/or M_y
- ✓ Shear forces: Q_y and/or Q_x



III/ COMBINED BENDING AND TENSION OR COMPRESSION (UỐN VÀ KÉO HOẶC NÉN) – Normal stress

- Normal stress due to normal force and bending moments:



$$\sigma_z = \frac{N_z}{A} + \frac{M_x}{I_x} y + \frac{M_y}{I_y} x$$

or

$$\sigma_z = \pm \frac{|N_z|}{A} \pm \frac{|M_x|}{I_x} |y| \pm \frac{|M_y|}{I_y} |x|$$

III/ COMBINED BENDING AND TENSION OR COMPRESSION (UỐN VÀ KÉO HOẶC NÉN) – Normal stress

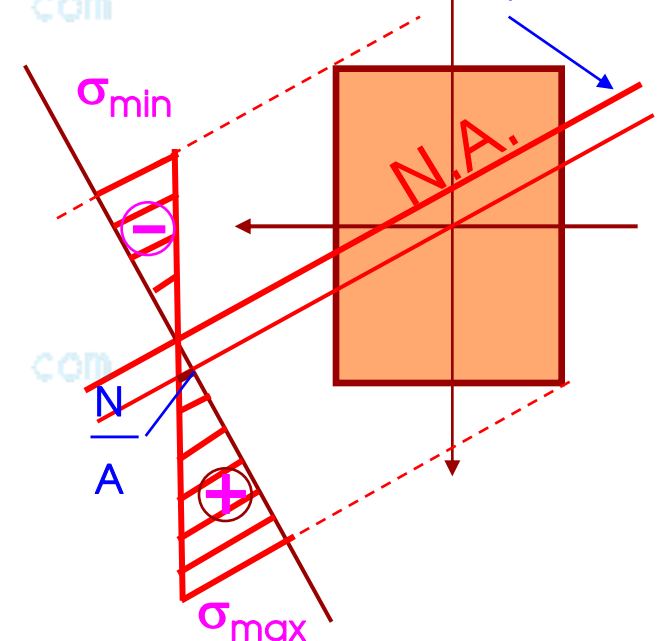
■ Neutral Axis – Normal Stress Diagram

✓ Biaxially symmetric sections

$$\frac{N_z}{A} + \frac{M_x}{I_x} y + \frac{M_y}{I_y} x = 0$$

$$\sigma_{\max} = \pm \frac{|N_z|}{A} + \frac{|M_x|}{W_x} + \frac{|M_y|}{W_y}$$

$$\sigma_{\min} = \pm \frac{|N_z|}{A} - \frac{|M_x|}{W_x} - \frac{|M_y|}{W_y}$$

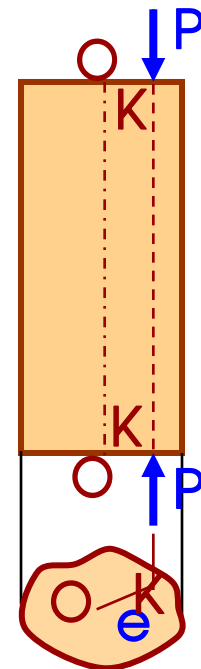


III/ COMBINED BENDING AND TENSION OR COMPRESSION (UỐN VÀ KÉO HOẶC NÉN) – Eccentric tension

- **Special case:** Eccentric tension (or compression) (kéo hay nén lệch tâm)

✓ When the bar is acted upon by two equal and opposite forces P which act along KK parallel to its axis

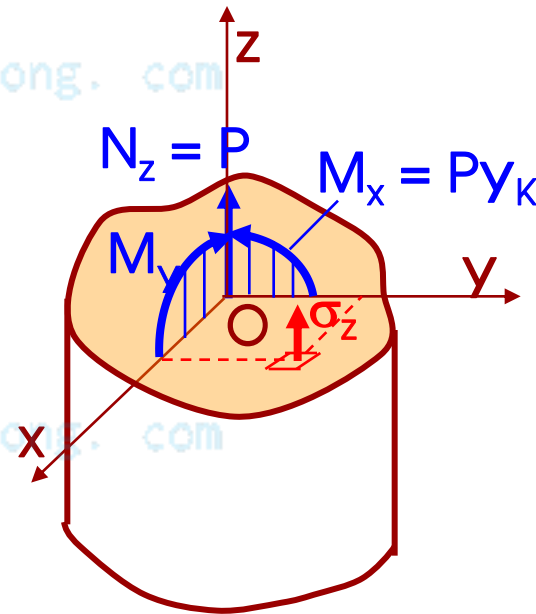
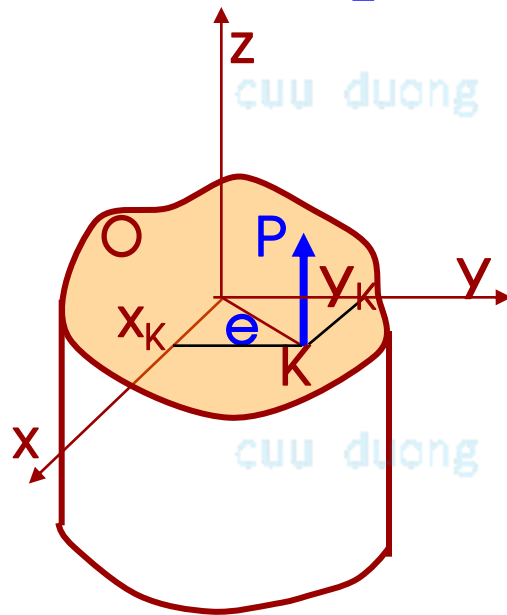
✓ $OK = e \rightarrow$ eccentricity



III/ COMBINED BENDING AND TENSION OR COMPRESSION (UỐN VÀ KÉO HOẶC NÉN) – Eccentric tension

✓ Internal forces:

$$N_z = P; \quad M_x = Py_K; \quad M_y = Px_K$$



III/ COMBINED BENDING AND TENSION OR COMPRESSION (UỐN VÀ KÉO HOẶC NÉN) – Eccentric tension

✓ Neutral Axis Equation (P/t đi trung hòa)

$$\sigma_z = \frac{P}{A} + \frac{P y_k}{I_x} y + \frac{P x_k}{I_y} x = \frac{P}{A} \left(1 + \frac{y_k}{r_x^2} y + \frac{x_k}{r_y^2} x \right) = 0$$

with $I_x = r_x^2 A$; and $I_y = r_y^2 A$

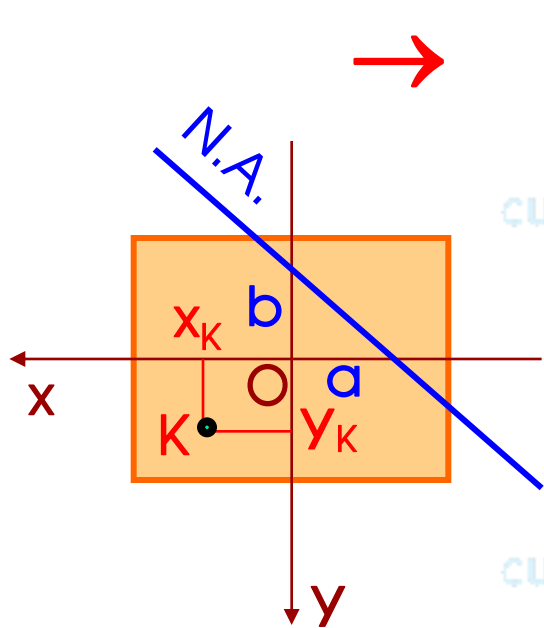
r_x, r_y – radii of gyration of section about the principal axes (bán kính quán tính)

III/ COMBINED BENDING AND TENSION OR COMPRESSION (UỐN VÀ KÉO HOẶC NÉN) – Eccentric tension

$$1 + \frac{y_K}{r_x^2} y + \frac{x_K}{r_y^2} x = 0 \quad \text{or} \quad \frac{x}{a} + \frac{y}{b} = 0$$

with $a = -\frac{r_y^2}{x_K}; \quad b = -\frac{r_x^2}{y_K}$

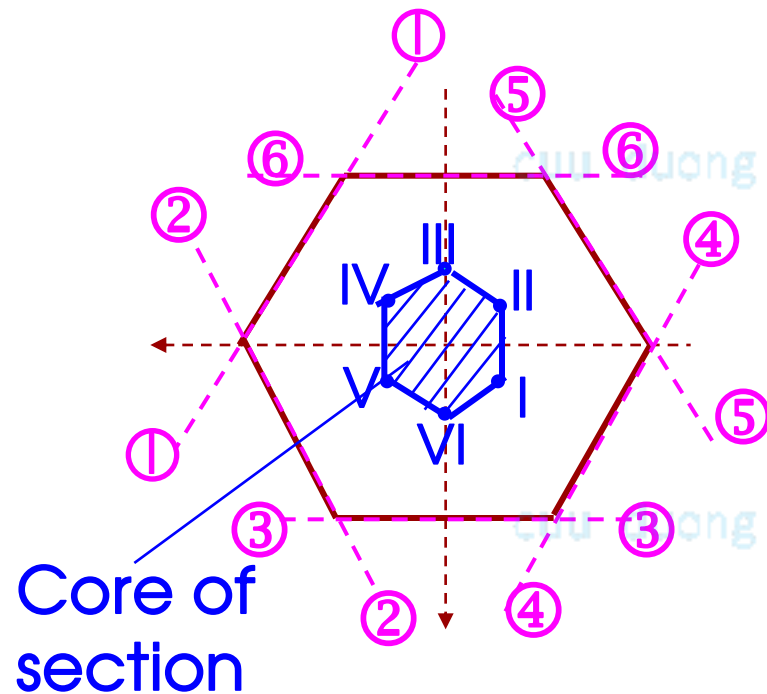
✓ Properties of neutral axis ?



The diagram shows a rectangular cross-section with a centroid O. A point K is located at coordinates (x_K, y_K) relative to O. The x and y axes are shown. The neutral axis (N.A.) is a line passing through K. The distances from O to the neutral axis are a and b.

III/ COMBINED BENDING AND TENSION OR COMPRESSION (UỐN VÀ KÉO HOẶC NÉN) – Core of section

■ Core of section (Lõi của tiết diện)



Ứng với 1 đường trung hòa $i-i$ (trùng với 1 cạnh của đa giác), tọa độ của điểm đặt lực I :

$$\begin{cases} x_I = -\frac{r_y^2}{a} \\ y_I = -\frac{r_y^2}{b} \end{cases};$$

$I = 1$, số cạnh lõi

III/ COMBINED BENDING AND TENSION OR COMPRESSION (UỐN VÀ KÉO HOẶC NÉN) – Core of section

✓ Core of a rectangular cross-section

• Neutral axis 1-1

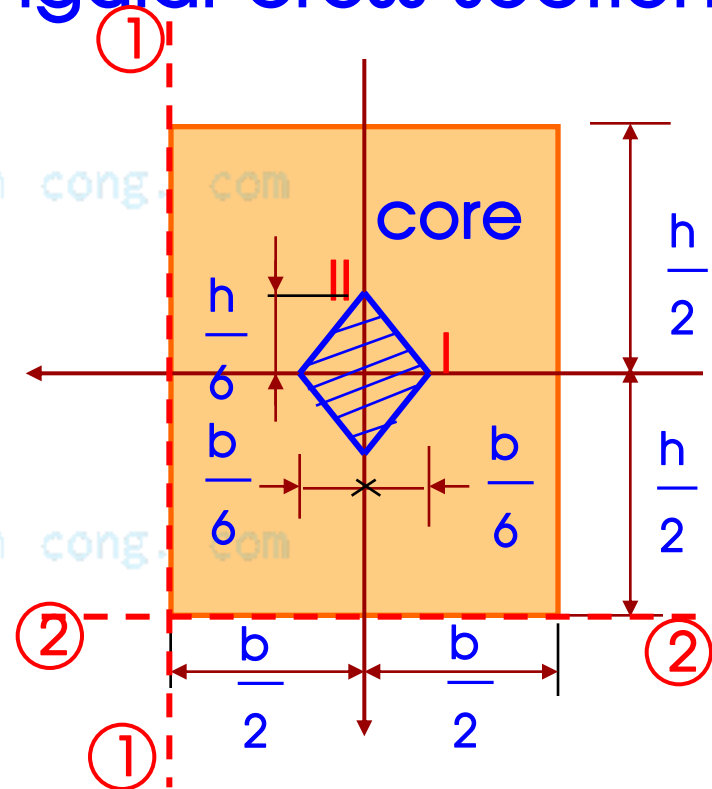
$$a = \frac{b}{2}; \quad b = \infty$$

$$I \left(x_I = -\frac{b}{6}; y_I = 0 \right)$$

• Neutral axis 2-2

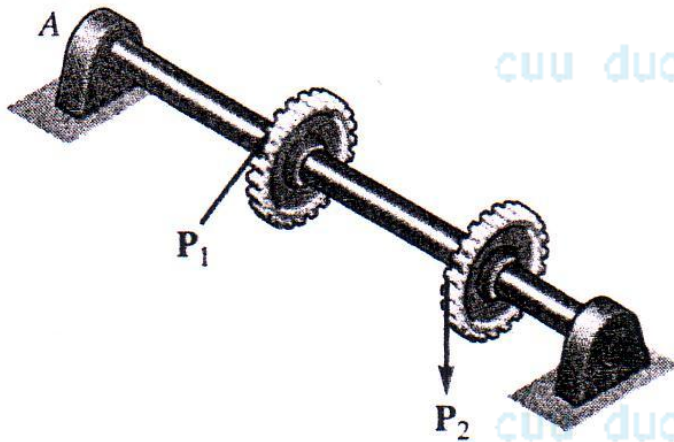
$$a = \infty; \quad b = \frac{h}{2}$$

$$II \left(x_{II} = 0; y_{II} = -\frac{h}{6} \right)$$

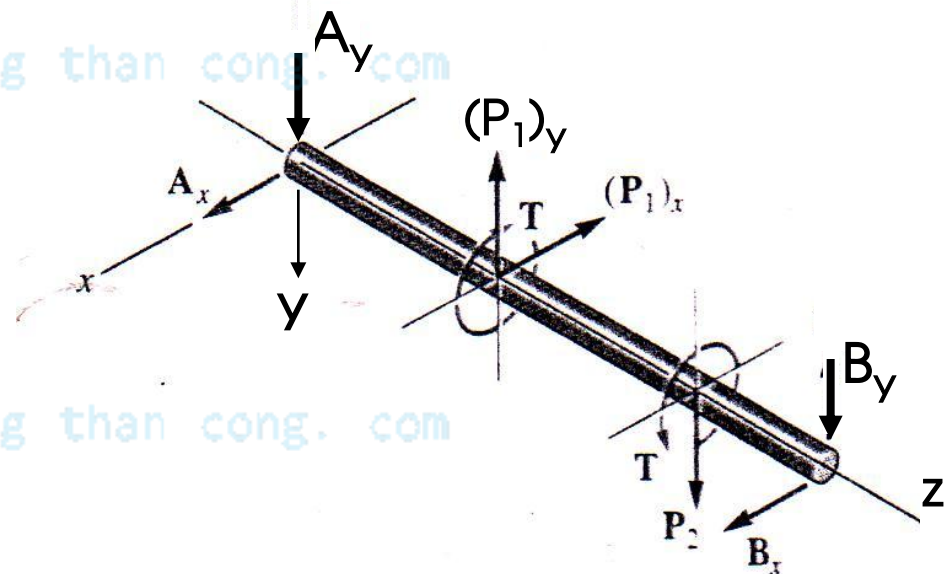


IV/ COMBINED BENDING AND TORSION (UỐN VÀ XOẮN ĐỒNG THỜI)

■ Introduction: Circular Shaft and pulleys



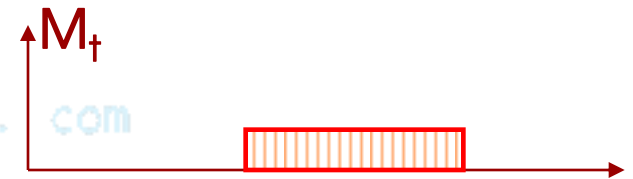
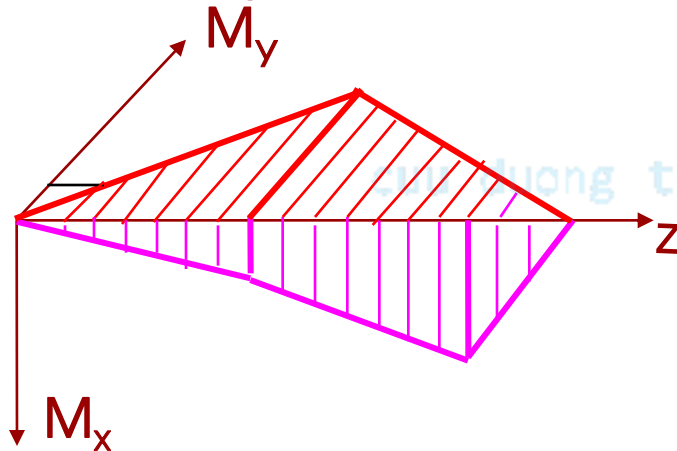
(a) Circular shaft and pulleys



(b) Free body diagram

IV/ COMBINED BENDING AND TORSION (UỐN VÀ XOẮN ĐỒNG THỜI)

Bending moment in x-z plane

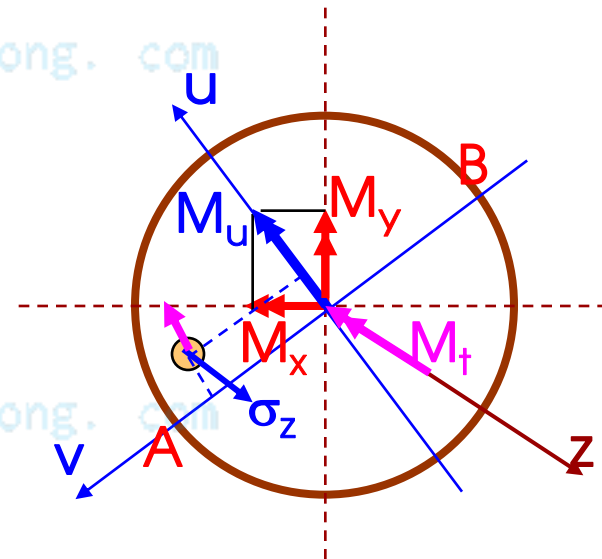
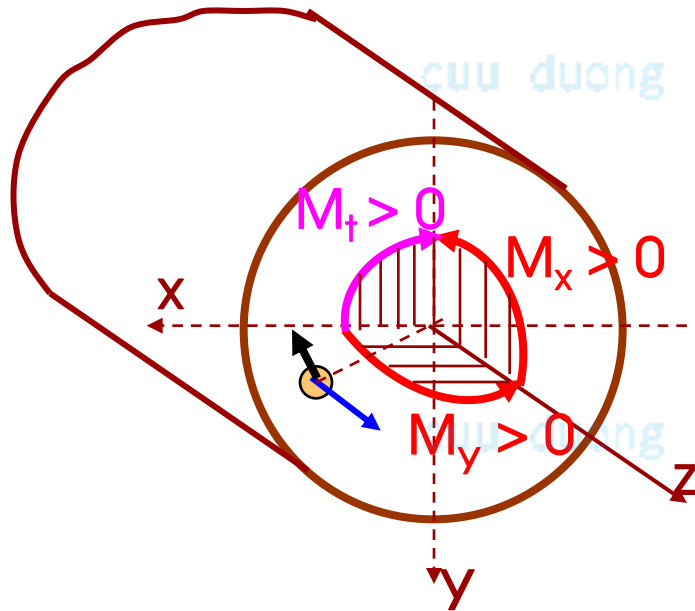


Bending moment in y-z plane

Torsion Moment about z-axis

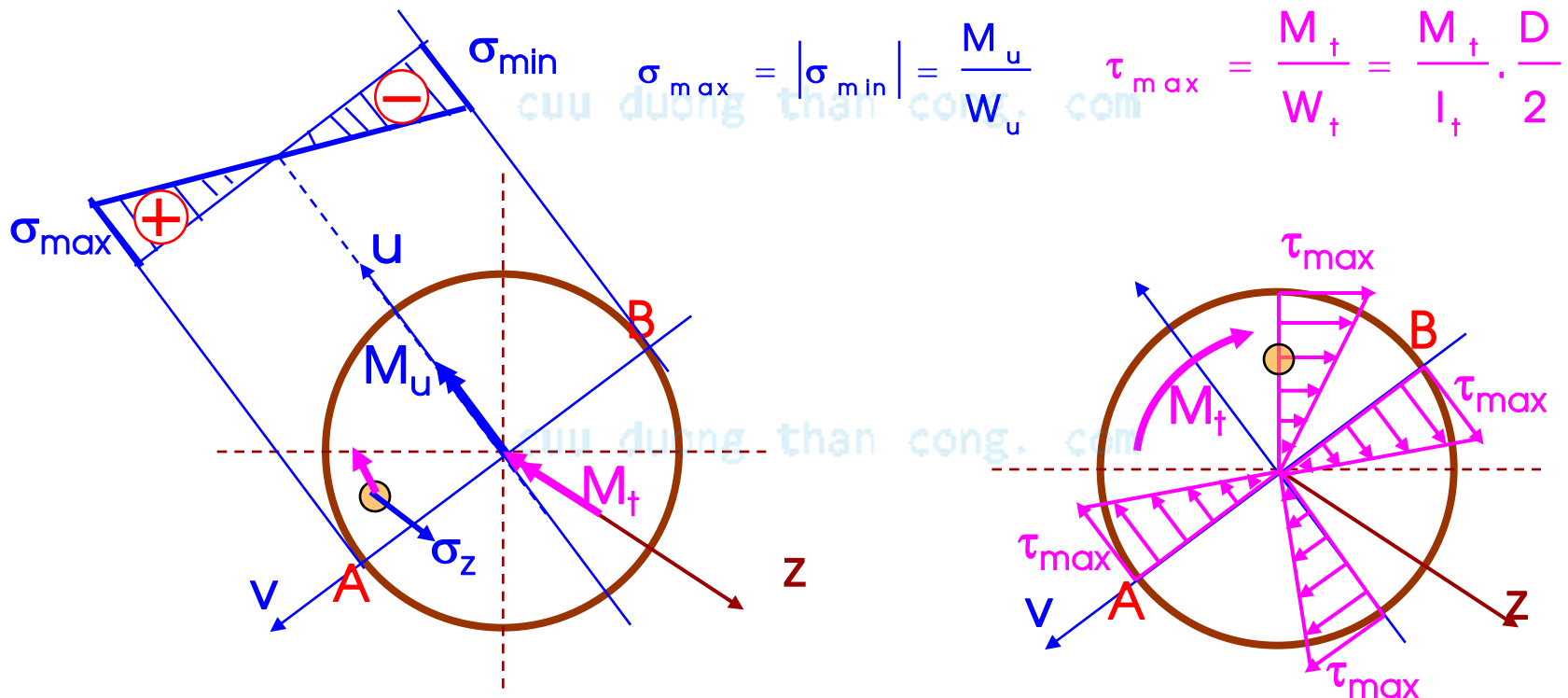
IV/ COMBINED BENDING AND TORSION (UỐN VÀ XOẮN ĐỒNG THỜI)

- Moment Components and Sign convention



IV/ COMBINED BENDING AND TORSION (UỐN VÀ XOẮN ĐỒNG THỜI)

■ Stresses distribution due to moments



IV/ COMBINED BENDING AND TORSION (UỐN VÀ XOẮN ĐỒNG THỜI)

- Strength condition

✓ Critical point: Point A

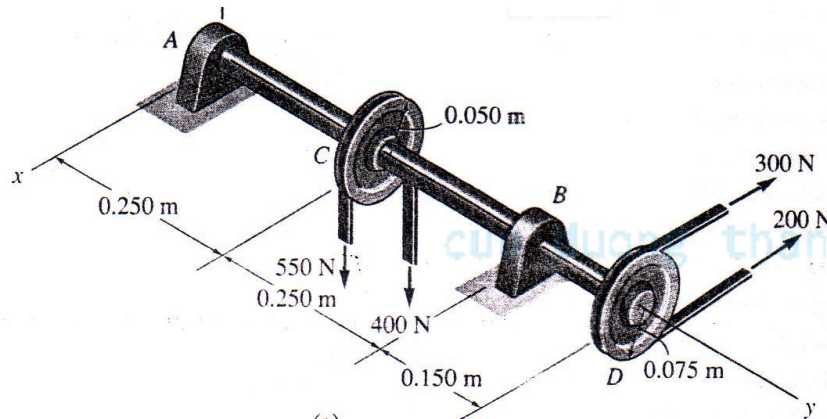
$$\sigma_A = \sigma_{\max} = \frac{M_u}{W_u} = \frac{\sqrt{M_x^2 + M_y^2}}{W_u}$$

$$\tau_A = \tau_{\max} = \frac{M_t}{W_t} = \frac{M_t}{2W_u}$$

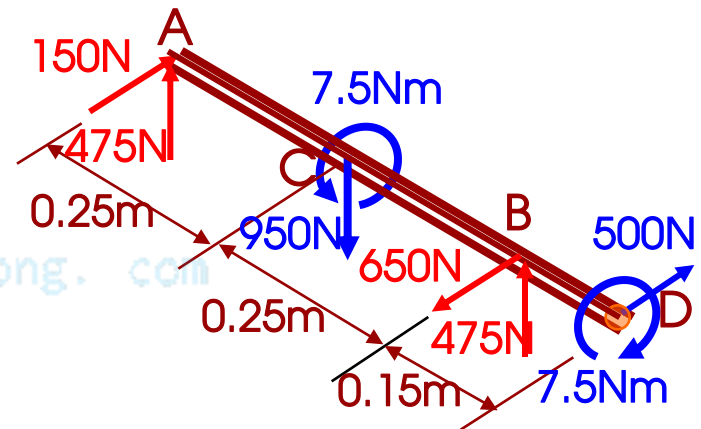
$$\rightarrow \sigma_{t3} = \sqrt{(\sigma_A)^2 + (\tau_A)^2} \leq [\sigma]$$

IV/ COMBINED BENDING AND TORSION (UỐN VÀ XOẮN ĐỒNG THỜI)

- Example:** The shaft in fig.a is supported by smooth journal bearings at A and B. Determine the smallest diameter of the shaft using the maximum-shear-stress theory with $\tau_{\text{allow}} = 50 \text{ MPa}$



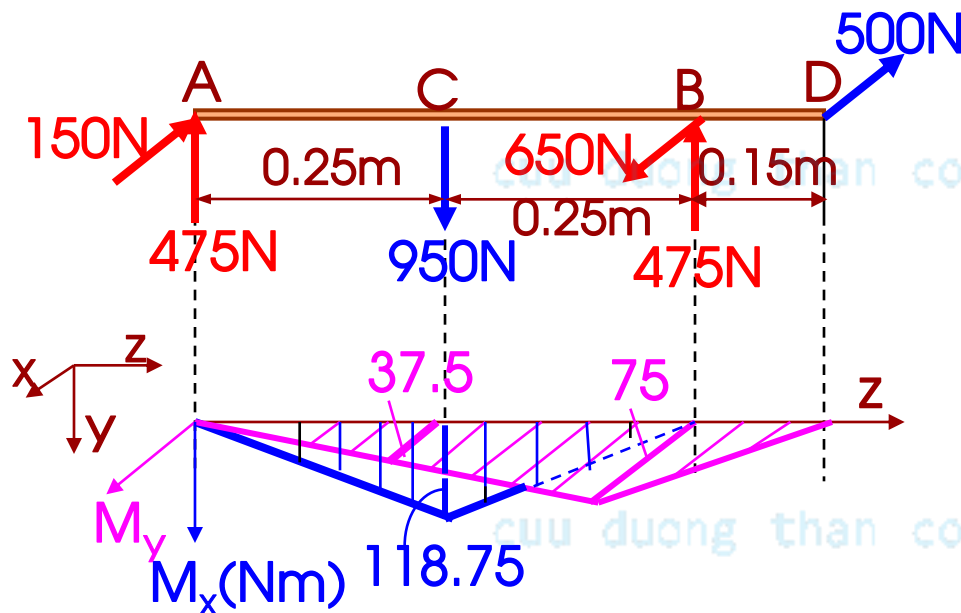
(a) Real scheme



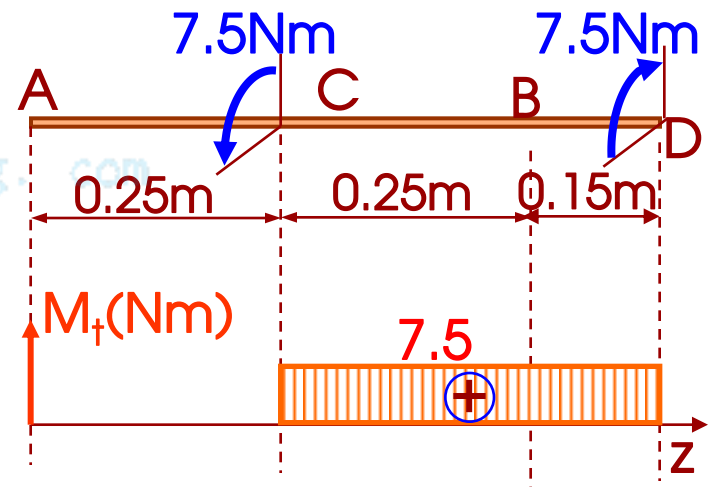
(b) Free-body diagram

IV/ COMBINED BENDING AND TORSION (UỐN VÀ XOẮN ĐỒNG THỜI)

■ Solution

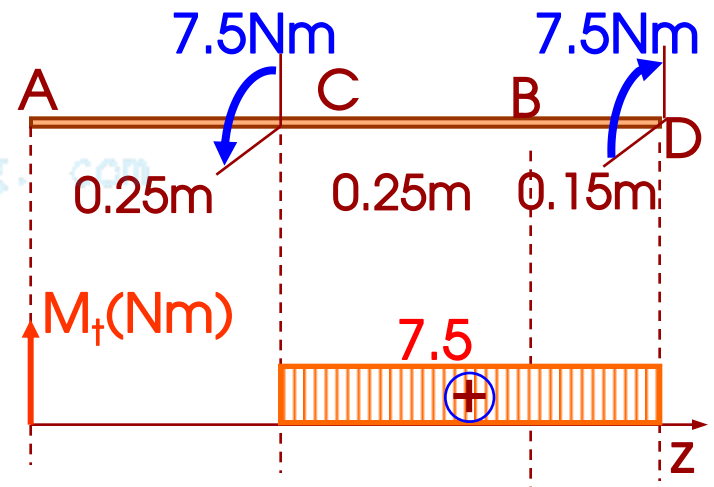
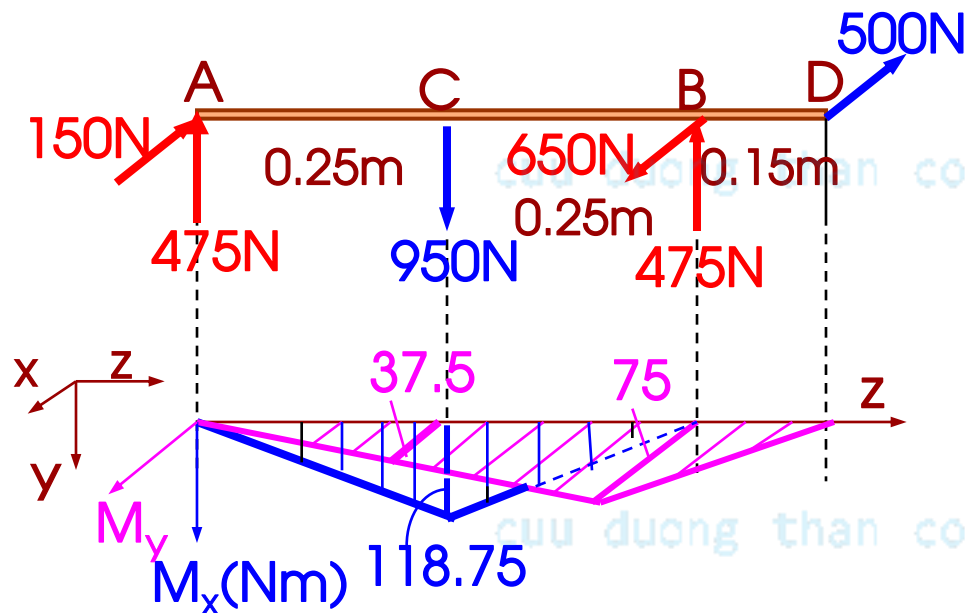


Bending moment diagrams



Torsion moment diagram

IV/ COMBINED BENDING AND TORSION (UỐN VÀ XOẮN ĐỒNG THỜI)



IV/ COMBINED BENDING AND TORSION (UỐN VÀ XOẮN ĐỒNG THỜI)

■ Critical point: C $\left\{ \begin{array}{l} |M_x| = 118.75 \text{ Nm} ; |M_y| = 37.5 \text{ Nm} \\ |M_t| = 7.5 \text{ Nm} \end{array} \right.$

$$M_u = \sqrt{M_x^2 + M_y^2} = \sqrt{(118.75)^2 + (37.5)^2} = 124.53 \text{ Nm}$$

$$\sigma_{t3} = \sqrt{\sigma^2 + 4\tau^2} = \frac{1}{W_u} \sqrt{M_u^2 + M_t^2} \leq 2\tau_{\text{allow}}$$

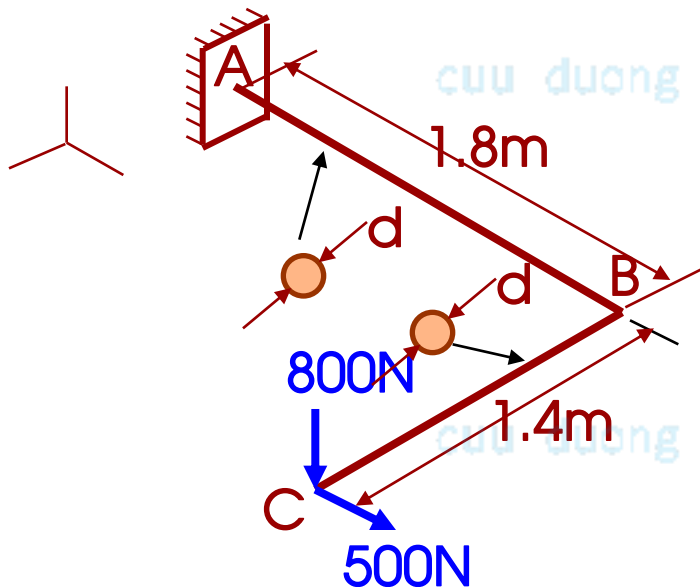
$$\rightarrow W_u = \frac{\pi}{32} d^3 \geq \frac{\sqrt{M_u^2 + M_t^2}}{2\tau_{\text{allow}}} = \frac{\sqrt{(124.53)^2 + (7.5)^2}}{2(50 \times 10^6)} = 124.75 \times 10^{-8}$$

$$d \geq \sqrt[3]{\frac{32 \times 124.75 \times 10^{-8}}{\pi}} = 0.0233 \text{ m}$$

Answer: $d = 23.3 \text{ mm}$

V/ GENERAL COMBINED LOADING

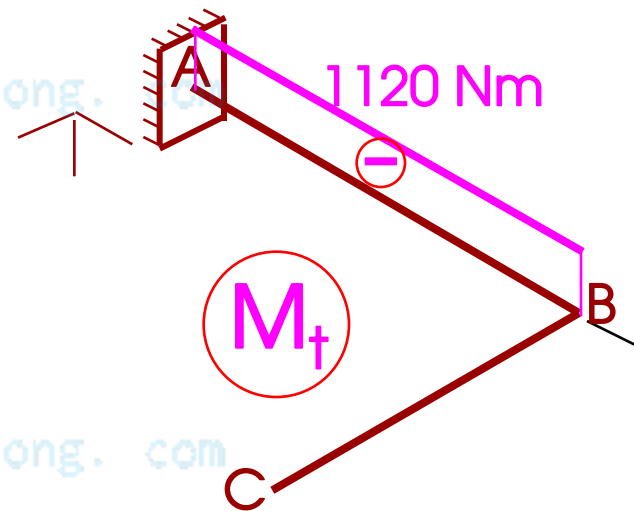
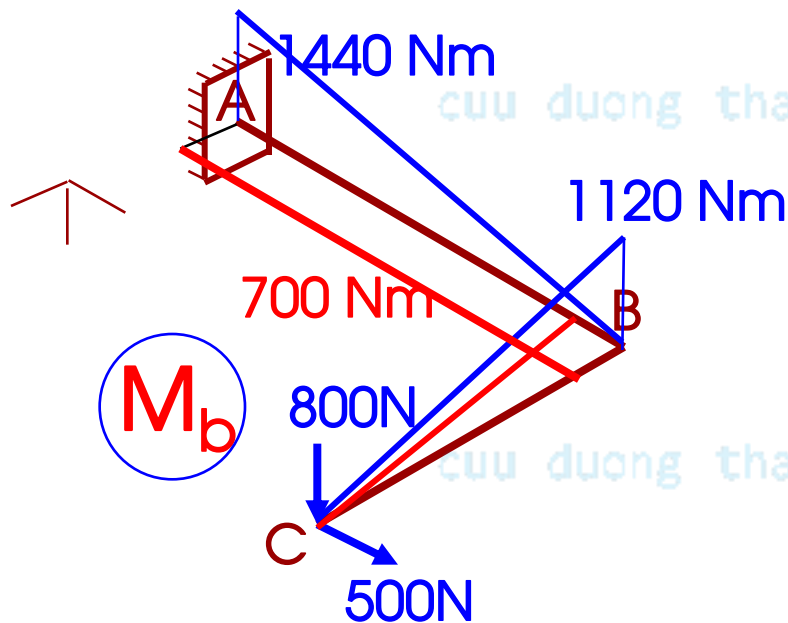
Example



- The solid rod ABC has a radius of $d = 50\text{mm}$. Verify the strength condition using the maximum-shear-stress theory if $[\sigma] = 160\text{MPa}$

V/ GENERAL COMBINED LOADING – (continued)

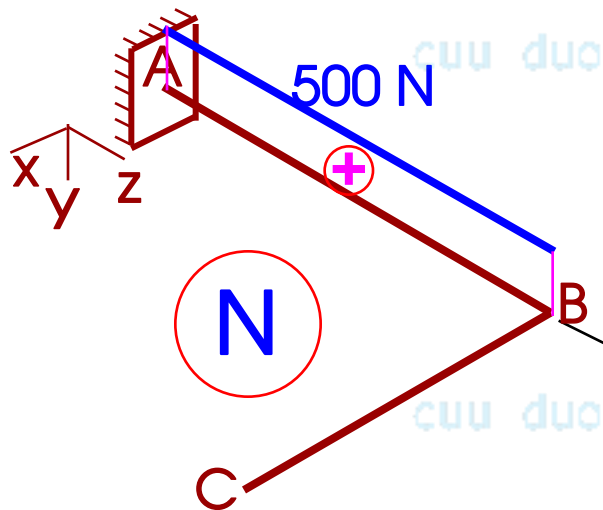
■ SOLUTION



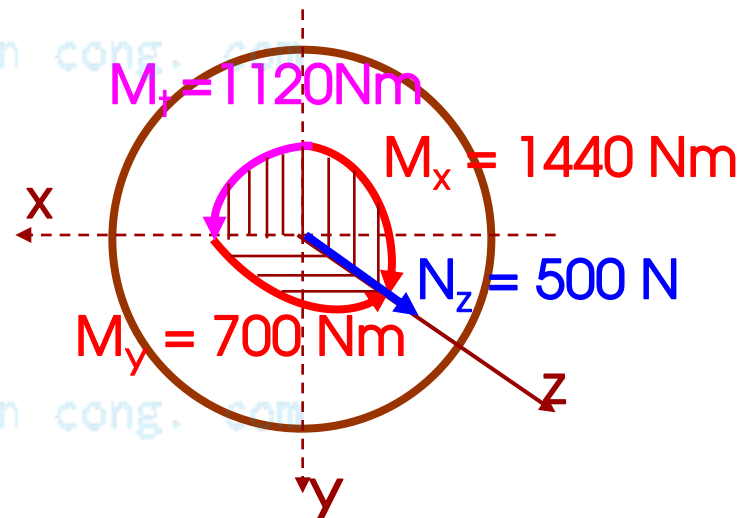
(a) Bending moment diagrams (b) Torsion moment diagram

V/ GENERAL COMBINED LOADING – (continued)

- Internal forces at built-up section A (Nội lực tại tiết diện ngàm A)

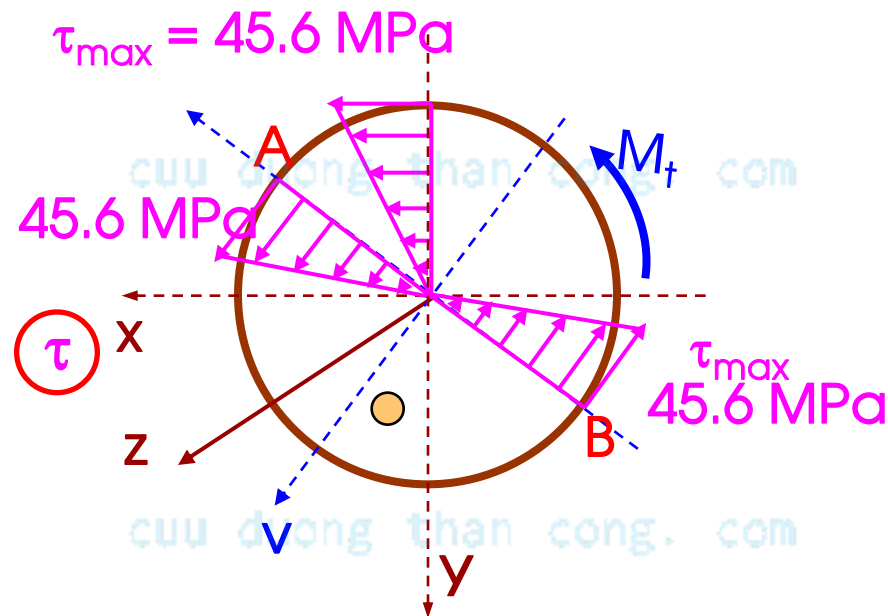


(c) Normal force diagram



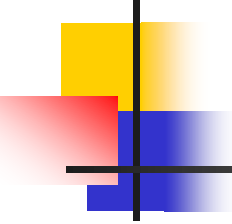
(d) Internal forces at section A

V/ GENERAL COMBINED LOADING – (continued)



Shear Stress Distribution





V/ GENERAL COMBINED LOADING – (continued)

✓Point A

$$\sigma^{(N)} = \frac{N}{A} = 0.25 \text{ MPa}$$

$$\sigma_A^{(M)} = \sigma_{\max}^{(Mu)} = \frac{M_u}{W_u} = 109 \text{ MPa}$$

$$\sigma_A = \sigma^{(N)} + \sigma_A^{(M)} = 0.25 + 109 = 109.25 \text{ MPa}$$

$$\tau_A = \tau_{\max} = \frac{M_t}{W_p} = 45.6 \text{ MPa}$$

$$\sigma_{t3} = \sqrt{(\sigma_A)^2 + 4(\tau_A)^2} = 142.3 \text{ MPa} \leq [\sigma] = 160 \text{ MPa}$$